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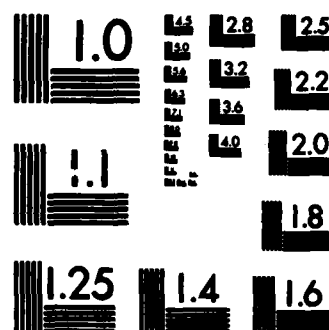
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USAF BIOENVIRONMENTAL NOISE DATA HANDBOOK

Volume 157

KC-10A IN-FLIGHT CREW NOISE

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AIR FORCE AEROSPACE MEDICAL RESEARCH LABORATORY
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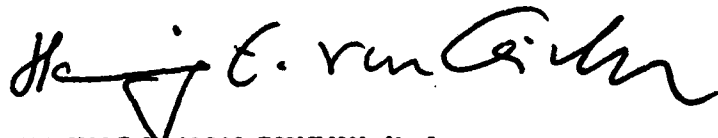
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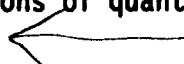
This technical report has been reviewed and is approved for publication.

FOR THE COMMANDER



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20. ABSTRACT (Continue on reverse side if necessary and identify by block number) → The KC-10A is the standard USAF tanker-transport aircraft with high-speed, high-altitude refueling and long range transport capability. This report provides measured data defining the bioacoustic environments at flight crew/passenger locations inside this helicopter during normal flight operations. Data are reported for 24 locations in a wide variety of physical and psychoacoustic measures: overall and band sound pressure levels, C-weighted and A-weighted sound levels, preferred speech interference level, perceived noise levels,		

and limiting times for total daily exposure of personnel with and without standard Air Force ear protectors. Refer to Volume 1 of this handbook, "USAF Bioenvironmental Noise Data Handbook, Vol 1: Organization, Content and Application," AMRL-TR-75-50(1) 1975, for discussion of the objective and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations, applications, limitations, etc. 

PREFACE

This report was prepared by the Biodynamic Environment Branch, Air Force Aerospace Medical Research Laboratory, under Project/Task 723109, Communication and Performance Capability and Operational Noises. The author acknowledges the efforts of Mr. John Cole who established the data analysis requirements, Mr. Henry Mohlman, and Mr. Fred Lampley of the University of Dayton who assisted in the mechanics of data processing and Mrs. Norma Peachey who typed this report and prepared it for publication.

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INTRODUCTION

The KC-10A is a USAF tanker-transport aircraft with high-speed high-altitude refueling and long range transport capability. This aircraft, which is manufactured by the McDonnell Douglas Corporation, is powered by three CF-6-50C1 turbofan engines each rated at 52,500 lbs. maximum takeoff thrust. The engines are manufactured by the General Electric Company, Aircraft Engine Group.

This volume provides measured and extrapolated data defining bioacoustic environments produced inside this helicopter. Such data are essential to evaluate ear protection requirements, limiting personnel exposure times, voice communication capabilities, and annoyance problems associated with operations of the KC-10A aircraft.

This volume is one of a series published by the Air Force Aerospace Medical Research Laboratory (AFAMRL) under the same report number (AMRL-TR-75-50) as a multi-volume handbook that quantifies the noise environments produced at flight/ground crew locations and in surrounding communities by operations of Air Force aircraft and ground support equipment. The far-field, community-type, noise data in the handbook describe the noise produced during ground operations of aircraft, ground support equipment, and other ground-based equipment or facilities.

Volume 1 of this handbook discusses the objectives and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing. Refer to Volume 1 (reference 1) for such information because it is not repeated in other handbook volumes.

A cumulative index lists those aerospace systems contained in the handbook, and identifies the specific volumes containing each type of environmental noise data available (i.e., inflight/flight crew and passenger noise, near-field ground crew noise, far-field/community noise). Volume numbers are assigned sequentially as individual volumes are published.

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1. Cole, John N., USAF Bioenvironmental Noise Data Handbook, Volume 1: Organization, Content and Application, AMRL-TR-75-50(1), Aerospace Medical Research Laboratory, Wright-Patterson Air Force Base, Ohio, 1975.

IN-FLIGHT NOISE

MEASUREMENTS

All noise measurements were made on-board a standard-configured KC-10A aircraft during typical speed, altitude, and flight maneuver conditions. These levels describe the standard KC-10A environments, but may not be representative of those levels encountered if the aircraft has been configured differently (e.g., major equipment or structural changes).

Acoustic measurements were made at various flight crew and passenger locations. Table 1 lists the measurement locations and test conditions as numeric/alphabetic designators which are used on the data pages. The designator 1/A means measurement location 1 and test condition A.

The microphone position was at ear level external to headgear in a region 0.2-0.3 meter from the head when an individual was present. At unoccupied locations, measurements were made at ear level throughout a volume where the head would normally be located. In both cases, the microphone was randomly moved throughout a spherical volume approximately 0.3 meter in diameter and the resultant samples analyzed using a 4- or 8- second integration time to obtain a power-averaged level, which effectively smooths out short-duration fluctuations and best describes the exposure.

Although the presence of a crew member or passenger at a measurement location affects the resultant sound field, the magnitude of such effects is generally small and not significant in determining exposure limits or voice communication capabilities. Consequently, no distinction is made in this report between occupied and unoccupied measurement locations.

RESULTS

The measured data presented in Table 2 define the sound pressure levels (SPL) produced inside the KC-10A aircraft at the 24 specified locations. This table includes the overall, 1/3 octave band, and octave band levels. From these data, C-weighted and A-weighted sound levels, maximum permissible time for one exposure per day (AFR 161-35) with and without standard Air Force ear protectors, preferred speech interference level, and perceived noise level are calculated and presented in Table 3. These measures are widely used to assess the effects of noise on personnel and their performance.

TABLE 1**MEASUREMENT LOCATIONS AND TEST CONDITIONS**

KC-10A, Barksdale AFB, 21 June 1982

LOCATION	POSITION	HEIGHT ABOVE DECK
1	Cockpit Between Pilot & Copilot	Seated Head Level
2	Cockpit Navigator/Flight Engineer	Seated Head Level
3	Station 480, Seat at Right Sidewall	40" Above Deck
4	Station 480, Seat at Centerline	40" Above Deck
5	Station 480, Seat at Left Sidewall	40" Above Deck
6	Top Bunk, Right Side, Curtain Closed	40" Above Deck
7	Top Bunk, Left Side, Curtain Closed	40" Above Deck
8	Bottom Bunk, Left Side, Curtain Closed	40" Above Deck
9	Bottom Bunk, Left Side, Curtain Closed	40" Above Deck
10	Station 1238, Centerline	40" Above Deck
11	Station 1238, Left Sidewall	40" Above Deck
12	Station 1238, Right Sidewall	40" Above Deck
13	Station 1538, Right Sidewall	40" Above Deck
14	Station 1538, Centerline	40" Above Deck
15	Station 1538, Left Sidewall	40" Above Deck
16	Station 1738, Left Sidewall	40" Above Deck
17	Station 1738, Centerline	40" Above Deck
18	Station 1738, Right Sidewall	40" Above Deck
19	Boom Operating Area, Middle Seat	Seated Head Level
20	Boom Operating Area, Pumps On, Middle Seat	Seated Head Level
21	Boom Operating Area, Pumps On, Left Seat	Seated Head Level
22	Boom Operating Area, Pumps On, Right Seat	Seated Head Level
23	Boom Operating Area, Retracting Boom, Doors Open	Seated Head Level
24	Boom Operating Area, Doors Closed	Seated Head Level

CONDITION DESCRIPTION

A	Engine Start
B	Taxi
C	All Engines Idle, Ground Runup
D	Takeoff/Roll
E	Climb to 5,000 ft.
F	Climb to 10,000 ft.
G	Climb to 20,000 ft. - 320 KIAS
H	Cruise - 37,000 ft. 0.8M
	AC - Automatic Mode With Both Pads Set on Auto
I	Cruise - 29,000 ft. - 325 KIAS
	Approach KC-135, Light Door Closed
J	Cruise - 29,000 ft. - Receiving Fuel From KC-135
K	Cruise - 29,000 ft. - 325 KIAS
	Approach KC-135, Light Door Open
L	Descent to 20,000 ft. - 320 KIAS
M	Approach - 3,000 ft. - 150 KIAS
N	Final Approach - 1,000 ft., Gear Down
P	Landing/Roll

TABLE: MEASURED SOUND PRESSURE LEVEL (DB)										IDENTIFICATION:	
2											
										OMEGA 3.2	
										TEST NR-082-001	
										RUN 01	
										02 AUG 82	
										PAGE F1	

TABLE: MEASURED SOUND PRESSURE LEVEL (DB)													IDENTIFICATION:	
2														
NOISE SOURCE/SUBJECT:														
KC-10A IN-FLIGHT													OMEGA 3.2	
CREW NOISE													TEST BR-082-001	
													RUN 02	
													02 AUG 82	
													PAGE F2	

TABLE: MEASURED SOUND PRESSURE LEVEL (DB)										IDENTIFICATION:	
2										OMEGA 3.2	
										TEST BR-082-001	
										RUN 03	
										02 AUG 82	
										PAGE F3	
NOISE SOURCE/SUBJECT: (OPERATION:											
KC-10A IN-FLIGHT											
CREW NOISE											
24/H 1/I 1/J 1/K 1/L 1/M 1/N 1/O 1/P										LOCATION/CONDITION	
FREQ (HZ)											
25	75	82	80	96	87	90	85	85	83		
31.5	78	77	75	78	77	83	82	78	80		
40	75	80	76	80	79	82	80	77	79		
50	79	77	76	78	75	82	78	73	78		
63	77	76	75	77	73	78	76	67	75		
80	76	78	74	80	77	80	78	66	74		
100	76	78	76	80	78	79	79	66	73		
125	80	81	83	84	73	78	76	64	73		
160	80	83	82	84	66	74	73	61	71		
200	81	73	79	81	67	71	71	56	67		
250	79	69	77	80	66	71	69	57	64		
315	78	70	77	78	65	68	68	57	61		
400	77	68	77	77	68	68	65	62	59		
500	75	69	77	76	69	66	63	69	58		
630	76	70	77	78	68	68	65	67	62		
800	72	75	74	77	70	70	69	58	59		
1000	68	70	72	76	68	67	65	54	55		
1250	67	70	72	75	68	67	65	55	56		
1600	67	67	71	73	65	65	62	56	57		
2000	65	62	68	71	63	62	60	54	55		
2500	62	59	65	69	59	59	56	51	56		
3150	60	57	62	64	56	55	54	55	53		
4000	56	57	57	60	53	52	52	47	51		
5000	54	51	54	56	51	49	49	40	45		
6300	52	50	52	54	50	50	47	41	43		
8000	52	50	53	54	48	47	46	40	42		
10000	51	48	52	53	46	45	44	40	42		
OVERALL	90	90	90	93	89	93	90	87	87		

LEVEL CORRECTED TO REMOVE BACKGROUND/ELECTRONIC NOISE.

TABLE: MEASURED SOUND PRESSURE LEVEL (DB)														IDENTIFICATION:	
2 OCTAVE BAND															
NOISE SOURCE/SUBJECT: (OPERATION:)														OMEGA 3.2	
KC-10A IN-FLIGHT ()														TEST BR-082-001	
CREW NOISE ()														RUN 01	
()														02 AUG 82	
()														PAGE J1	
LOCATION/CONDITION															
FREQ (HZ)	1/A	1/B	1/C	1/D	1/E	1/F	1/G	1/H	2/H	3/H	4/H	5/H	6/H	7/H	8/H
31.5	70	80	80	96	88	85	85	85	84	81	76	80	80	80	80
63	68	74	73	91	81	81	81	81	80	79	73	78	78	80	82
125	67	74	73	87	77	81	80	80	76	75	75	76	76	77	79
250	63	70	70	81	73	72	72	69	69	72	72	72	71	72	73
500	60	65	65	74	69	73	74	72	75	72	76	73	71	73	73
1000	63	68	67	73	68	74	75	72	71	74	74	74	71	75	74
2000	59	64	63	69	66	68	71	67	66	71	73	72	66	68	68
4000	53	59	58	64	63	60	62	58	57	61	64	62	56	57	59
8000	46	51	50	58	62	55	59	51	51	56	61	62	51	52	63
OVERALL	75	83	82	98	89	89	88	87	86	85	83	85	84	85	86

TABLE: MEASURED SOUND PRESSURE LEVEL (DB)		IDENTIFICATION:														
2	OCTAVE BAND															
NOISE SOURCE/SUBJECT:		OMEGA 3.2														
KC-10A IN-FLIGHT		TEST BR-082-001														
CREW NOISE		RUN 02														
		02 AUG 82														
		PAGE J2														

TABLE: MEASURED SOUND PRESSURE LEVEL (DB)		IDENTIFICATION:	
OCTAVE BAND			
2			OMEGA 3.2
			TEST BR-092-001
			RUN 03
			02 AUG 82
			PAGE J3

NOISE SOURCE/SUBJECT:		OPERATION:		LOCATION/CONDITION	
KC-10A IN-FLIGHT					
CREW NOISE					

FREQ (HZ)	24/H	1/I	1/J	1/K	1/L	1/M	1/N	1/O	1/P
31.5	81	84	82	87	88	91	98	86	86
63	82	82	80	83	80	85	82	74	81
125	84	86	86	88	79	82	81	69	77
250	84	76	82	85	71	75	74	61	69
500	81	74	81	82	73	72	69	71	65
1000	74	77	77	81	73	73	71	61	62
2000	70	68	73	76	68	67	65	59	61
4000	62	60	63	66	58	57	57	56	56
8000	56	54	57	58	53	52	51	45	47
OVERALL	90	90	90	93	89	93	90	87	87

TABLE: MEASURES OF HUMAN NOISE EXPOSURE													IDENTIFICATION:	
3													OMEGA 3.2	
NOISE SOURCE/SUBJECT: (OPERATION:)													TEST BR-002-001	
KC-10A IN-FLIGHT ()													RUN 01	
CREW NOISE ()													02 AUG 82	
()													PAGE M1	
()														
LOCATION/CONDITION														
1/A	1/B	1/C	1/D	1/E	1/F	1/G	1/H	2/H	3/H	4/H	5/H	6/H	7/H	8/H
HAZARD/PROTECTION														
C-WEIGHTED OVERALL SOUND LEVEL (OASLC IN DBC) AT EAR														
A-WEIGHTED OVERALL SOUND LEVEL (OASLA IN DBA) AT EAR														
MAXIMUM PERMISSIBLE TIME (T IN MINUTES) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73)														
NO PROTECTION														
OASLC	73	81	80	95	87	88	85	85	84	82	83	83	84	85
OASLA	86	71	71	79	74	77	78	75	77	79	78	74	77	77
T	960	960	960	960	960	960	960	960	960	960	960	960	960	960
H-157 IN-FLIGHT COMMUNICATION UNIT														
OASLA*	50	56	56	70	60	64	62	61	59	59	59	59	60	62
T	960	960	960	960	960	960	960	960	960	960	960	960	960	960
COMMUNICATION														
PREFERRED SPEECH INTERFERENCE LEVEL (PSIL IN DB)														
PSIL	61	66	65	72	68	72	73	70	71	72	74	73	69	72
ANNNOYANCE														
PERCEIVED NOISE LEVEL, TONE CORRECTED (PNLT IN PNDB)														
TONE CORRECTION (C IN DB)														
PNLT	80	86	85	95	89	91	91	89	90	90	93	92	86	89
C	2	1	2	1	1	1	1	1	2	0	1	1	0	0

* BASED ON CALCULATED SPL SPECTRUM UNDER PROTECTIVE DEVICE.

TABLE: MEASURES OF HUMAN NOISE EXPOSURE															IDENTIFICATION:		
3															OMEGA 3.2		
															TEST BR-082-001		
NOISE SOURCE/SUBJECT:															RUN 02		
KC-10A IN-FLIGHT																	
CREW NOISE															02 AUG 82		
															PAGE M2		

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2. BASED ON CALCULATED SPL SPECTRUM UNDER PROTECTIVE DEVICE